# **Design Example 9** Cantilevered Monotube Support for a Dynamic Message Sign

## **Problem Statement:**

Design a cantileved monotube structure in Ft. Collins, CO. It will support a dynamic message sign weighing 5,000 pounds. Assume a 24" diameter circular tube fabricated from A36 steel. Bolts are ASTMA 325 bolts. The structure would cross a lifeline travelway on failure.



# **Overview Monotube Design Example**

- Video 1 Example Introduction & Strength Loads
- Video 2 Tube Strength Design Checks
- Video 3 Base Plate Design
- Video 4 Fatigue Loads
- Video 5 Fatigue Resistance and Design Checks

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Date:	Class:	Assignment:
WIND	LOADS EWIND	IN & DIRECTION CRITICAL]
TUBE #1	BEHIND SISN -	NO WIND AREA
TUBE #2	$A_{452} = \frac{2\pi(10)}{4} \times 2 =$	31.42 fte EIGNORE LOSING PORTION BERIND SIS
TUBE #3 ,	Aw3 = 10 (2) = 204	Cy 2
WIND P	RESSURE PZ = 0.00	0256 K2 K2 G V C2
STRUCTUR	RE CROSSES LIFELIN	NE TRAVEZOWAY TABLE 3.8-1
3.8.4 Kz=	$2.0\left(\frac{2}{2g}\right)^{2/2}$ $2g = 900$ Exp.	RISK CATEGORY = HIGH USE 1700yr MRE GUNO C V=120mph
Conse	ERVATIVELY USE 2=	24 St FOR ALL HEIGHTS
	K2 = 2.0 (24) 2/9.5	5 = 0.98
3.8.6 9	+ = 1.14	
3.8.5 TA	BCE 3.8.5 - 1 Kg	= 0.85 (DYN, MSB SISN : HORIZ ARM SPA
3.8.7 Cd	FACTOR TABLE .	3.8.7-1
	CPLINDRICAL	WITH V.d > 78 mph G= 0.45
	DYNAMIC I	MESSAGE SIGN Cd = 1,70
TR	RESSURE P2 = 0.0025	56 (0.93) (0.85) (1.14) (120) <sup>2</sup> Cd
		= 33.22 Cd
	FOR TUBE Pa=	33.22 (0.45) = 14.95 pst
	FOR SISN P2 =	33.22 (1.70) = 56.48psf

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Date:	Class:	Assignment:
WIND FOR	CES	
TUBE #1	W, = O (BEN)	IND SISN )
TUBE #2	W2 = 31.42 ft 2 (1	14.95pst) = 469.716
TUBE #3	W3 = 20 (14	.95) = 299.016
SIGN	$W_{5} = (8 \times 24) 50$	6.48 = 1084416
Nomina	L WIND LOA	O EFFECTS
At SPLI	CE	
1 <sup>y</sup>	V2 = 10844	4" = 10.84" (SHEAR)
×	My = 1089	14(8) = 86.75 ft.k (BENDING)
AT BA	ISE	
1Y	× Vz = 10844	+ 469.7 + 299 = 11.61 K (SHEAR)
1	Mx = 10844(:	20)+469,7(16.37)+299(5) = 226.1 ft.k (BERUDING)
E	My = 10844(1	18)+469.7(3.63) = 196.9 ft.k (TORSION)
FACTORE	ED LOADING	EXTREME I LIMIT STATE
TAB	LE 3.4-1 8	= 1.25
	80	= 1.10 + dw = 1.0 - WILL CONTROL SD=0.9 FOR DPLIFT WILL NOT

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Date:		Class:	Assi	gnment:	
AT SOL	INE				
Mr Orb	Non LOAD	D	W		FACTORED 1.1D+1.0W
AXMAL	$\mathcal{P}_{x}$	0	0		0
SHEAR	VY	6.93K	0	AY	7.62%
SHEAR	$\bigvee_{z}$	0	10.84 %	X	10,84K
TORSION	mx	15 ft.k	0	2	16.5 ft.k
BENDING	Μy	0	86.75 ft.1c		86.75 A.K
BENDING	Ma	58,93ft.k	0		64.82 ft.16
Ŧ	TOTAL	SHEAR ACTORED 7	$V = \sqrt{Vy^2 + V}$	$\overline{I_{2}^{2}} = 13.$	25 K
DES					
Des	-	AXIAL Pu	= 0		
Des		AXIAL Pu SHEAR Vu	= 0 _= 13.25K		
Des		AXIAL Pu SHEAR Vu BENDING	= 0 1= 13.25K Mu = 108.3f	+·K	

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Name:

Date: Class: Assignment: AT BASE NOMINAL FACTORED W D 1.1D+1.0W LOAD SHEAR Vx O 0 0 AXIAL Py 11.30 (c) 0 12.434 SHEAR V2 0 11.61K 11.61% BENOING My 15.0 4.16 226.1 A.K. 242.6 ft.K Torsion My 0 196.9 FA.K 196.9 Ft.K BENDING M2 137.6A.2 0 151.4 ft.k TOTAL BENDING M= VMx2+M02 = 286.0 ft.k TOTAL SHEAR V = V + V2 = 11.61" DESIGN FACTORED FORCE EFFECTS AXIAL Pu= 12.43" (C) SHEAR Vu = 11.61K BENDING Mu = 286.0 ft. K TORSION Tu = 196.9 H.K.

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JUBE TO SPLICE 24"9× 3/8	DESISN CHE	CKS FLEXURE $P_{p} = 0.90$ SHEAR $P_{v} = 0.90$ AXIAL COMP $P_{e} = 0.90$ TORSION $P_{T} = 0.95$
<u>SPUCE</u> 24"9 × 3/8	Peo Vela	SHEAR $\phi_r = 0.90$ AXIAL COMP $\phi_r = 0.90$ TORSION $\phi_r = 0.95$
<u>SPLICE</u> 24"9 x 3/8	DEA Vala	AXIAL COMP PE = 0.90 TORSION PT = 0.95
<u>SPUCE</u> 24"9 x 3/8	PER VIII	TORSION \$7=0.95
<u>SPLICE</u> 24"9× 3/8	P=0 V - 13	K ind
24"q x 3/8		25" M. = 108 3 M. 2 T - 11. 5 52. 4
24"px 3/8		
	R= 11.81" +=	3/0 "
	C.	A= 6.28 Rt = 27.82 2
	ADDENDIN	$B = 7 = 2140^3 \pm = 1940 \text{ in}^4$
	TIPPEIDUR	SE - 1.27 SHAPE FARTOR
		5 = 21482 + = 144 7 53
		C1 = 6 22 024 = 292 5 in 3 TODATON CONST
		-frendr - Jadis in Store of
BENDING 2	4 = 24/ Na = 64 > 1	= 0 07 E = 0.07 (29000) = 5C AGT COMPACT
	e nerel / n	Fu 36
	TABLE 5.7.2-16	Last = 250 los Dhel
	- /	1 = 0.31 E = 200 1 E
TADIE 582	-1	NON-COMPACT
Ance 010.5	ma - ma	001/0(E/E)] - 097 MA.
	1 = 0,7	7 + 0.000 My = 0.17 Mp
	6	
n	p = SF + SF = 1.2	$7(164.2h^3)(36ksic)(\frac{144}{12in}) = 625.6$ ft.k
	$M_n = 0.97  m$	$n_p = 606.8  \text{ft.k}$
	$M_r = \varphi_r M_n =$	0.9 (606.8) = 546.2 ft.k
SHEAR	Vn= AvFnv A	$v = A_g = 2\frac{7.82}{2} = 1391iv^2$
,		
F	1.60E	0.73E BUT
inv - max 1	JIV/D13/4 MAG	(D/+) 12 THAN OF GTY
6	VB(Z)	OR EQUAL TO
Ly Dis	TANCE TO ZERO SHE	SAR V LV = 20 ++
1	1.10(29000)	0.78(29000)
F = mAx (T	= BIKSL A	ND = 44.2ks. ( ± 0.6(36) = 21.6143
100 3.1	74 (64) 14	(64)"+ ( A
(~	<i>•</i> -1	2
		Fmy = 21.6145
	Vn = 13.91 m2 6:	21.6ksi) = 300.4 K
	V = 0	01/=09(300.4)=270.4/k

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Date: Class: Assignment: TORSION  $T_n = C_t F_{nt}$   $C_t = 328.5 in^3$  $\overline{F_{ht}} = m_{AX} \left\{ \frac{1.23E}{\sqrt{\frac{L}{D}}} \left( \frac{D_{|t|}}{2} \right)^{5/4} = 62 \text{ ksi} \text{ AND } \frac{0.6E}{(D_{|t|})^{3/2}} = 34 \text{ ksi} \right\} \le 0.6F_{Y} = 21.6 \text{ ksi}$ Fn+= 21.6ksi Th = 328.5 in 3 (21.6 ksi) (15+) = 591.3 ft.k Tr= O.Tn=0.95 (591.3) = 561.7 ft.k COMBINED FORCES INTERACTION EQUATION Tu = 16.5 = 0.03 < 0.2 TORSION & SHEAR CAN Tr 561.7 BE IGNORED  $\frac{T_{u}=0}{T_{r}} = \frac{T_{u}}{2T_{r}} + \frac{T_{mu}}{M_{r}} = \frac{108.3}{546.2} = 0.20 \times 1$ SPLICE SECTIONS OK. BASE P= 12.43k(c) Vu= 11.61k Mu= 286.0 Ft.K Tu= 19/6.9 Ft.K 24" px"/16 R= 11.66" t="/16 A = 6.28 Rt = 50.34 in2 APPENDIX B I= 3.14R3 t = 3422.1 in4 SF = 1.27 SHAPE FACTOR 5 = 3.14R2 = 293.5 in3 C1 = 6.28 R2 E = 587.0 in3 BENDING DH= 24/16 = 34.9 e 2p=56 COMPACT Mn = Mp = SF \* SFy = 1.27(293.5)(36)(12) = 1118.24.k Mr = \$\$\$ Mn = 0.9 (1118.2) = 1006.4 ft.k

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Date:	Class:	Assignment:
SHEAR	Vn = Ar For	$\begin{array}{c} A_{V} = A_{g} = 5_{0.34} = 25.17 \text{ in}^{2} \\ 2 & 2 \end{array}$
		For = 0.6 Fy = 21.6 ksi = Plt LOWER THAN 24x3/8 TUBE
	$V_{\rm M} = 25.1$	7(21.6) = 543.7 K
	$V_r = \Phi_V V_r$	= 0.9 (345.1) = 981.5
TORSION	Th = Ct Fnt	G=587.0in3 Fnz=0.6Fy=21.6ksi
	$T_n = 587.$	0(21.6)(12) = 1056.674.1K
	Tr= PrTn=	0,95(1056.6) = 1003.8 A.K
AXIAL		
COMPRESSION	Pnc = Ag F	F = APPENDIX B F = 0.707R = 8.24''
	Ag = 50.34	$\tilde{w}^2$ USE L = 20++ <u>KL</u> = <u>2.1(20 × 12)</u> = G1.2 K=2.1 <u>F</u> <u>8.24</u>
KL 2 4.1	1) E = 134 Fy	$\frac{D}{E} = 34.9 < \lambda_r = 0.11E = 88$
	$\overline{F}_{e} = \underline{\Pi^{2}E}$	= 76.4 KSL
	$(^{\mu}4_{r})^{2}$	$(\frac{5}{76})$ $(\frac{36}{76.4})$
		For = 0.658 Fy = 0.658 (36) = 29.56 KSL
	Pnc = 50.	34 (29.56) = 1487.9 K
	Pre = Pel	Rc = 0.9 (1487.9) = 1339.0K

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Class: Assignment: Date: COMBINED FORCES WTERACTION EGN  $\frac{T_{u}}{T_{r}} = \frac{196.9}{1003.9} = 0.197 \ 20.20 \ T_{ORSION} \ SHEAR \\ CAN BE IGNORED$  $\frac{P_{u}}{P_{u}} = \frac{12.43}{1339} = 0.01 < 0.2$  $\frac{P_u}{2R_r} + \frac{B_m u}{r} = \frac{12.43}{2(1339)} + \frac{1003(286)}{1006.4} = 0.30 \times 1$ BASE SECTION OK  $B = \frac{1}{1 - \frac{P_u}{P_0}} = \frac{1}{1 - \frac{P_u}{A_0 F_c}} = \frac{1}{1 - \frac{12.43}{50.34(76.4)}} = 1.003$ EXAMPLE \* IF TU > 0.2 (ALMOST FOR THIS CASE)  $\left(\frac{P_{u}}{P}\right) + \left(\frac{B_{mu}}{m_{r}}\right) + \left(\frac{V_{u}}{V_{r}} + \frac{T_{u}}{T_{r}}\right)^{2}$  $= \frac{12.43}{1339} + \frac{1.003(286)}{10064} + \left[ \left( \frac{11.61}{4893} \right) + \left( \frac{196.9}{10020} \right) \right]^{2} = 0.35$ POLE TO BASE PLATE WELD 1/2" FILLET WELD, ETO ELECTRODES P= 12.43" V= 11.61" BMu=1003(286)=2869 File ANALYZE PER 1" OF WELD Tu = 19609 Ft.K AXIAL BA = P = 12.43 = 0.17 K/im SHEAR  $g_{V} = \frac{V_{u}}{A_{12}} = \frac{11.61}{75.4/2} = 0.31^{k/in}$ CIRCLE  $A_{V} = \frac{A_{0}}{7}$ r=12" A= 2111 = 75.4 in I= Mr3 = 5429 in 3 3 = TTr2 = 452.4 n2 J=217r3 = 10857 in3 BENDING &m = BMU = 286.9×12 = 7.61 K/in TORSION & = TF = 196.9(12)(12) = 2.61 K/in

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Date: Class: Assignment: COMBINE FORCES ON WELD (VECTORALLY)  $g = \left(\frac{T^{2}}{A} + \frac{BMu}{5}\right)^{2} + \left(\frac{V}{A/2} + \frac{T_{F}}{T}\right)^{2} = \left(0.17 + 7.61\right)^{2} + \left(0.31 + 2.61\right)^{2} = 8.31^{10} \text{ Min}$ WELD CAPACITY (ASSUME SHEAR ON WELD THROAT) Qw = 0.75 Fnw = 0.6FEXx = 0.6(70) = 42ksi PRn = Qw Fnw (0.707 Ew) = 0.75 (42ksi) (0.707 × 1/2) = 11.14 1/1 7 8 WELD OK. THRU-THICKNESS CHECK PR= Ptever 0.6Fu= 0.75 ("/16) 0.6 (58ksiv) = 17.9 K/in O.K. THROAT THRU THICKNESS

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UPPER NUT

BASE

PLATE

OWER (LEVELING)

NUT

CIVIL AND ARCHITECTURAL ENGINEERING Class: Assignment: Date: BASE PLATE DESISA P= 12.43" (C) V= 11.61" BM. = 286.9 A.K. T= 196.9 A.K. 40" 
 base plate 24" \u00f3 tube 32" φ bolt pattern 16.551230 30° USE AISC DESIGN GUIDE | BASE PLATE AND ANCHOR ROD DESIGN STRUCTURAL DOUBLE NUT CONNECTION SUPPORT ANCHOR TTACHMENT) BOLT ANCHOR BOW GROUP PROPERTES WASHER 11/2" \$ F1554 Fy= 55Kin F= 75Kin EMBEDMENT ANALYZE IN FORCE PER BOLT n=12 Figure C5.16-1-Typical Double-Nut Connection I= 2 [2(16.5=11.30) =+ 2(16.5=1166) =+ (16.5) ]

 $S = \frac{I}{16.5} = 99 in$   $T = 2r^2 = 12(16.5)^2 = 3267 in^2$ 

I = 1633.5 in2

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Date: Class: Assignment: ANCHOR FORCES AXIAL Fp = 12.43 = 1.04 " COMPRESSION SHEAR Fy = 11.61 = 0.97 K BENDING Fm = BM = 2869 (12) = 34.78 " TORSION F= Tr = 196.9(12)(165)= 11.93 K COMBINE ACTIONS ON ANCHORS ANCHOR AXIAL = Fp + Fm = 1.04+34.78 = 35.82K ANDARN SHEAR = F. + F. = 0.97 + 11.93 = 12.90 K ASSUME NO BENDING IN ANCHOR FROM SHEAR ACCORDING TO CURRENT LTS-6 5.17.3.1 IF CLEAR DISTANCE IS LESS THAN ONE ANCHOR DIAMETER. ASSUME NO ANCHOR AXIAL FORCE INCREASE DUE TO PRYING ACTION ACCORDING TO CURRENT LTS-6 IF BASE PLATE THICKNESS = ANCHOR BOLT DIAMETER AISC DESIGN GUIDE | USES AISC STEEL SPECIFICATIONS CONSIDER TENSION SHEAR INTERACTION  $A_{b} = \pi \frac{(1-1)^{2}}{4} = 1.767 \text{ in}^{2}$ fa = 35.82 = 20,27 ksi fy = 12.90 = 7.30 ksi AISC TABLE J3.2 SHEAR QROV= \$ (0.563Fu) AD \$ = 0.75 Fu= 75ks PFny= p(0.563 Fu) = 31.67 Ksi PROV = 55.96 K > Vu= 12.90 K OR @Fny = 31.67 Ksi > fy = 7.30 ksu ANCHOR SHEAR OK.

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Date:	Class:	Assignment:
AXIAL	PRnt = PFnt'	4b 9=0.75
	REDU	UCED TENSION STRESS DUE TO
		TENSION SHEAR INTERACTION
	Fnt' = 1.3 Fnt -	- Ent for & Fint
		PFNV WHERE FAL= 0.75 Fu = 56.25 ksi
	Fne' = 1.3(56.25)	- 56.25 (7.30) = 60ksi No REDUCTION
		31.61 ". Fit = Fit = 56.25Ksc
	9 Rnz = 0.75 (5	6.25)1.767 = 74.55 × 74 = 35.82 K
	OR	
	PFnE' = 0.75	(56.25) = 42. 19 ksi > fa = 20.27 ksi
		ANCHOR AXIAL OK.
$MOMENT$ $M_{A} = 12.9$ $6.45ink$	IN ANCHOR 90" (1/2")= 6.45 in. k 12.90 k 11" ALCORANCE	Bolt 13" 1 Nut Base Plate
12.90 K	I CLEAKA NCE	24" 4
6.451	n-k	1 inch between Bottom of nut and
R Sere	and Manual	Concrete Assume fixed rotation and double curvature
(ASSUMENG	FULL DIA)	12,48
S6= 17 (11/2 32	)3= 0.331 in 3	Foundation
ADDITIONA	AXIAL	
STRESS ?	DUE TO	645 inth - 10 day "
ANCHOR	BENDING TM =	0.331 in 3

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Date: Class: Assignment: RECHECK AXIAL DESIGN OF ANCHOR WHEN CONSIDERING ANCHOR BENDING PFn1' = 42.19ksi > fat fm = 20.27+19.47= 39.74ksi ANCHOR BENGING DOUBLES AXIAL STRESS IN ANCHOR IMPACT OF ANCHOR BENDING NO ANCHOR BENDING WITH ANCHOR RENDING AXIAL = 20.27 Ksi AXIAL = 20.27 KSL BENDING = 0 BENDING = 19.47KSL @Fort' = 42.19ksi > 20.27ksi 9Fort'= 42.19ksi > 39.74ksi PERFORMANCE RATIO = 0.48 PERFORMANCE RATIO = 0.94 WILL IMPACT DESIGN

Name:

CURRENT LTS-6

#### 5.17.3.1—Double-Nut Anchor Bolt Connections

The design stresses on anchor bolts shall be determined in accordance with Article 5.17.4.1. In determining the compression effects, bearing of the base plate on concrete or grout shall be neglected. The allowable stresses for the anchor bolts shall be as determined in Article 5.17.4.2. Anchor bolts in double-nut connections should be pretensioned according to Article 5.17.5.

If the clear distance between the bottom of the bottom leveling nut and the top of concrete is less than the nominal anchor bolt diameter, bending of the anchor bolt from shear forces or torsion may be ignored. If the clear distance exceeds one bolt diameter, bending in the anchor shall be considered according to Article 5.17.4.3.

#### 5.17.4.3—Bending Stress in Anchor Bolts

When the clearance between the bottom of the leveling, nuts and the top of the concrete foundation exceeds one bolt diameter, bending stresses in the anchor bolts should be considered.

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The combined tension and shear and compression and shear requirements of Article 5.17.4.2 shall be used to account for the combination of bending, tension, compression and shear. Eqs. 5-24 and 5-25 shall be met with the value of  $f_i$  equal to the summation of the axial tensile stress and the maximum tensile bending stress or  $f_c$  equal to the summation of the axial the maximum tensile bending stress and the maximum tensile bending stress and the maximum tensile bending stress.

EXCEPTION NOT CURRENTLY IN LRFD SITS SPEC

IF CLEAR DISTANCE > ANCHOR DIA - INCLUDE BENDING

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Date: Class: Assignment: PRYING ACTION CONSIDERATION CURRENT LT3-6 5.17.3 DESIGN BASIS The axial force in anchor bolts that are subject to Prying effects of the base plate should be taken into consideration in the design strength of anchor bolt tension, or combined shear and tension, shall be calculated connections. However, research (NCHRP Report 412) has with consideration of the effects of the externally applied tensile force and any additional tension resulting from) shown that if the base plate thickness is equal to the anchor prying action produced by deformation of the base plate. bolt diameter, these prying effects may be neglected. EXCEPTION NOT CURRENTLY IN LRFD SLTS SPEL IF CHECKING PRYING ACTION ALSC STEEL MANUAL SECTION 9 (ASSUMING BASE PLATE MAY BE GROUTED) 2" BASE P=9" TRIB WIDTH 6=41/2" a=31/2 b=3.75 16.5 a=4.25 TRIB. WIDTH AXIAL + 8 THICKNESS REDUIRED SO THAT NO PRYING ACTION OCCURS B = BOLT TENSION = 35.82" (CONSERV.)  $t_e = \frac{4Bb'}{\sqrt{9pFu}}$ 9=0.90 Fu= 58ksi (A36 PLATE)  $t_{e} = \frac{4(35.82)(3.75'')}{10.9(9'')58k_{5}} = 1.07'' 2 t_{R} = 2''$ 50 NO PRYING ACTION

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Assignment: Date: Class: BASE PLATE BENDING AISC GUIDE / CANTILEVER APPROACH Dur qu 35.82× EFFEATUE 41/3" BENDING WIDTH M=35.82(42)=161.2 in. K BENDING SECTIONS 35.82 " 15 2"×9" A36 PLATE QMn = 0.9 (9in3) (36 ksi) = 291.6 in. K > 161.2 in. k PLATE BENDING OK. STRENGTH DESISN CHECK. SUMMARY LOADS: DEAD : WIND EXTREME I LIMIT STATE 1.115+1.0W STRENGTH: MONOTUBE AT SPUCE & BASE POLE TO BASE PLATE WELD BASE PLATE DESIGN

Name:

· ANCHORS

· PLATE BENDING

# **Design Example 9** Cantilevered Monotube Support for a Dynamic Message Sign

## **Problem Statement:**

Design a cantileved monotube structure in Ft. Collins, CO. It will support a dynamic message sign weighing 5,000 pounds. Assume a 24" diameter circular tube fabricated from A36 steel. Bolts are ASTMA 325 bolts. The structure would cross a lifeline travelway on failure.



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Date:	Class:	Assignment:
Fa	tique Checks	
/	Natural Wind	11.7.1.2
•	$T_F = 1$	Table 11.6-1
(	$C_d = 1.1$	Table 3.8.7-1 (pole)
P	$P_{NW} = 5.2  psf(2)$	TF)(Cd) 11.7.1.2-1
C	$p_{0} = 1.7$	Table 3.8.7-1 (sign)
FA	$iiisign = 5.2 (I_F)$	)(Cd) 11.7.1:2-1
PNH	isign = 5.2(1)(1.7	() = 8.89  psf.
PNA	PBle = 5.2(1)(1.1)	= 5.72 psf
FNA	kign = 8.84 (24'x	8') = 1.70 K
FNWE	bend = $5.72\left(\frac{2\pi r}{4}\right)$	(z') r=10'
Frank	bend = 0.180 K	
FNND	ole(bottom) = 5.72	$(10' \times 2') = 0.114^{k}$

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Date: Class: Assignment: 14.37' 201 0.18 1.7 5' 0.114 K 2.20 = 34.0<sup>1</sup>k  $M_{x} = 1.7(20)$ +0.18(16.37) = 2.95+ 0.114(5) = 0.57 37.5 1k

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Date:	Class:	Assignment:
Truck Cdsi Cdp	k Gusts gnvert = ole =	1. Z 1. I 1. I 1. I 1. I 1. I 1. I 1. I 1. I
PTO PTOS	=(18.8  psf)	$(I_F)(Cd) = 22.56 \text{ psf}$
PTGPS	$l_{e} = 18.2$	3(1)(1.1) = 20.68
FTGSig FTGPole	e = 22.5	$b(12'   ane \times 4') = 1.08^{k}$ $b(12' \times 2') = 0.50^{k}$
FTG-total MTG-total = =	= 1.58 = 2.3 <sup>k</sup> (24') = 37.8 <sup>ik</sup>	$ \begin{array}{c}                                     $

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Date: Class: Assignment: Check detail at top of stiffener 6.2 Tube-to-transverse plate  $K_F \le 2.5$ :  $K_I \leq 5.5:7.0$ In tube wall at the toe connections stiffened by 11.0 of the attachment to longitudinal attachments with tube weld at the Δσ partial- or full penetration termination of groove-welds, or fillet-welds in attachment. which the tube is subjected to longitudinal loading and the welds are wrapped around the (See detail In tube wall at the toe attachment termination. 5.4) (See detail of tube-to-transverse 5.4) plate weld. Fillet-welded tube-Weld toe on (11.9.3.1 - 14)to-transverse plate tube wall at the  $K_F = \left(\frac{t_{ST}^{0.4}}{t_T^{0.7}} + 0.3\right) \times \left(0.4 \times \frac{D_T^{0.8}}{N_{cm}^{1.2}} + 0.9\right)$ connections end of stiffened by attachment longitudinal Valid for: 0.25 in.  $\le t_{ST} \le 0.75$  in.;  $8 \le N_{ST}$ ; attachments  $0.25 \text{ in.} \le t_T \le 0.625 \text{ in.}; 24 \text{ in.} \le D_T \le 50 \text{ in.}$  $K_{F} = \left(\frac{0.5}{0.68^{0.7}} + 0.3\right) \left(0.4 \times \frac{24}{12^{1.2}} + 0.9\right)$ = (0.99+0.3)(0.26 + 0.9) = 1.29(1.16) = 1.50

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Date: Class: Assignment:  $K_{I} = \left[ (1.76 + 1.83(0.68)) - 4.76(0.22^{K_{F}}) \right]$ KE=1.5  $K_{I} = [1.76 + 1.24 - 0.49] 1.5$  $K_T = 2.51(1.5) = 3.76$ KI 5.5 .: AFTH = 7 Ksi.

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Date: Class: Assignment: Check Tube-to-Plate Detail 6.2 see pg.6

Table 11.9.3.1-2-Fatigue Stress Concentration Factors, KF

Section Type	Detail	Location	Fatigue Stress Concentration Factor for Finite Life, $K_F$	Section Type
	Fillet-welded tube- to-transverse plate connections	Fillet-weld toe on tube wall	$K_F = 2.2 + 4.6 \times (15 \times t_T + 2) \times (D_T^{1.2} - 10)$ $\times (C_{BC}^{0.03} - 1) \times t_{TP}^{-2.5}$ Valid for: 0.179 in. $\le t_T \le 0.5$ in.; 8 in. $\le D_T \le 50$ in.; 1.5 in. $\le t_{TP} \le 4$ in; 1.25 $\le C_{BC} \le 2.5$	(11.9.3.1–12)
Round	Groove-welded tube-to-transverse plate connections	Groove-weld toe on tube wall	$K_{F} = 1.35 + 16 \times (15 \times t_{T} + 1) \times (D_{T} - 5)$ $\times \left(\frac{C_{BC}}{4 \times C_{OP}}^{0.02} - 1}{4 \times C_{OP}}\right) \times t_{TP}^{-2}$ Valid for: 0.179 in. $\leq t_{T} \leq 0.625$ in.; 8 in $\leq D_{T} \leq 50$ in.; 1.5 in. $\leq t_{TP} \leq 4$ in.; 1.25 $\leq C_{BC} \leq 2.5$ ; 0.3 $\leq C_{OP} \leq 0.9$	(11.9.3.1–13)

Fillet-welded tube- to-transverse plate connections stiffened by longitudinal attachments	Fillet-weld toe on tube wall	$K_{F} = \left[ \left( 130 \times \frac{D_{T}^{0.15}}{N_{ST}^{1.5}} + 1 \right) \times \left( \frac{0.13}{h_{ST} + 7} \right) \times \left( \frac{6.5}{t_{ST}^{0.5}} - 1 \right) \right]$ × K <sub>F</sub> as per Equation (11-9.3.1-1) Valid for: 12 in. $\le h_{ST} \le 42$ in.; 0.25 in. $\le t_{ST} \le 0.75$ in.; $8 \le N_{ST}$ ; 24 in. $\le D_{T} \le 50$ in.	(11.9.3.1–1

Name: CIVIL AND ARCHITECTURAL ENGINEERING Date: Class: Assignment: KER ( without stiffener ) 11.9.3.1-13 KF0= 1.35 + 16 (15t, +1) × (D, -5)  $\frac{\times (C_{BC}^{0.03} - 1)}{(4C_{OP}^{-0.7} - 3)} t_{TP}^{-2.0}$ KF0= 1.35 + 16 (15(0.68) +1) (24-5)  $\left(\frac{1.375}{4(0.25)^{0.7}-3}\right)\left(2^{-2}\right)$  $K_{FO} = 1.35 + 179.2(19)(1.009599 - 1)(0.25)$ 7.556 KF0= 1.35 + 0.72 KED= 2.07

For stiffened connection 11.9.3.1-15

 $K_{F} = \left(\frac{130}{N_{T}} \frac{D_{T}}{N_{T}} + 1\right) \left(\frac{0.13}{h_{-} + 7}\right) \left(\frac{6.5}{+0.5} - 1\right) K_{FO}$  $K_{F} = \left[ \frac{130}{12^{1.5}} \right] + 1 \left[ \frac{0.13}{12+7} \right] \left[ \frac{6.5}{0.5^{0.5}} - 1 \right] \left[ K_{FO} \right]$ 

CIVIL AND ARCHITECTURAL ENGINEERING Date: Class: Assignment:  $K_F = \int 5.03 + 1 \int \int 0.00684 \int [8.192] [2.07]$ K= = 0.70 KI = [1.76 + 1.83 (0.68) ] - 4.76 (0.22) ] 0.70  $K_{I} = (3.00 - 1.64)(0.7)$  $K_{T} = 0.95$ · DFTH = 7 ksi 5 5.5 I, = 3732 in 4 As= 3.5 (0.5) = 1.75 in 2  $D_{\tau} = 24''$  $r_5 = \frac{24}{7} + \frac{3.25}{7} = 13.625"$ Js=12(1.75)(13.625) J= 3898 in +  $I_{s} = \frac{1}{2} = \frac{1949 \text{ in } 4}{2}$ ITotal = 3732 + 1949 = 568/ in4

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Date: Class: Assignment: Long Stiffener to Plate AFA = 10 ksi. Detail 6.3 Bolts - threads Defail 2.3 AFTH = 7 KSi 10.0 In base metal at Longitudinal stiffeners welded to 6.3 Transverse load-bearing 44.0 the weld toe or base plates. partial joint penetration groove-welded or filletthrough weld throat. welded attachments where t Δσ Δσ  $\leq 0.5$  in. and the main member is subjected to minimal axial and/or flexural loads (When t > 0.5 in, see note c).

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Date: Class: Assignment: Check Limit State At top of stitleners: AFTH = 7 ksi  $\Delta f = M_{faf} = \frac{72.4(12)}{303} = 2.9 \text{ ksi} \le 7 \text{ ksi}.$ Ratio = 2.9 = 0.41 OK At tube - to - plate: DF\_+ = 7 ksi  $\Delta f = M_{fat} = \frac{72.4(12)}{367} = 2.4 \quad Ratio = 0.34$ ok  $S_{bot} = \overline{I_{Total}} = \frac{5601}{15.5} = 367 \text{ in}^3$ At bolts  $\Delta F_{bolts} = \frac{M_{fat}}{99 \, A_{bolt}} = \frac{72.4(12)}{99 \, ih^2} = 8.5^{k}$ At with = 5.8 = 5.0 ksi & 7ksi. Ratio = 0.7/

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Date: Class: Assignment: Assume no stiffeners w CJP weld K== 2.08 K\_= [(1.76 + 1.83 t\_) - 4.76 (0.22 K\_F) | K\_F Kr= / (1.76 + 1.83 (0.68) - 4.76 (0.22 2.08) 7 2.08 K=(1.83)(2.08) = 3.8 K253->10 3. < K, × 4 -> 7.0 AFTH = 7.0 KSi. 4 < K = \$ 6.5 -> 4.5  $\Delta f = 72.4(12) = 2.87 \text{ ksi.}$ Ratio = 2.87 = 0.41 (Room for economy!)

CIVIL AND ARCHITECTURAL ENGINEERING Date: Class: Assignment: Try a smaller section. t = "/16 × 0.41 => 5/16 try 3/8"  $K_E = 1.35 + 16(15t_++1)(D_+-5)($  $\begin{pmatrix} C_{BC} - I \\ 4C_{C}^{-0.7} \\ 4C_{C}^{-0.7} \end{pmatrix} t_{TP}^{-2}$  $K_{E} = 1.35 + 16(15(0.375)+1)(24-5)$  $\binom{(1.375)^{0.02}}{4(0.25)^{0.7}-3}(2)$ KF= 1.35 + 16 (6.625) (19) (0.000846) (0.25) K== 1.35 + 0.426 = 1.78 K= 1.76 + 1.83 (0.375) - 4.76 (0.22 1.78) 1.78 KT = 2.12 \$ 3.0 AFTH = 10 KSi. Af = 72.4/12) = 5.27 < 10 ksi. 0k  $S_{\text{new}} = 303 \left( 0.375 \right) = 165 \text{ in}^3$  $Ratio = \frac{11/16}{0.375} = 1.83$ 11/16

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